

Asymptotical models of boundary layer flow control

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Abstract. Investigated are local flows in the laminar boundary layers in the vicinity of heating elements. On the basis of asymptotical analysis mathematical models are formulated and similarity parameters are found. Determined are flow parameters providing flow control (separation, transition). Presented are results of numerical and analytical analysis.

Another method of the flow control is associated with the technology of new materials development. For example, porous metals allow use of passive control methods to influence boundary layer separation or laminar-turbulent transition

Local surface heating. Among different methods of boundary layer flow control one of the mostly investigated now is a method associated with energy release due electrical discharge, surface heating or cooling.

The aim of this paper is the analysis of possible application of the local surface heating to determine response of the boundary layer flow and to find optimal heating elements parameters.

It is supposed that on the surface of the body are located heated parts, having temperatures different from the gas temperature in ambient boundary layer flow. It is supposed that temperature difference may change with time. Practically this method is easy to fulfill using electrically conducting strips. Example of such method application is described for example in [1].

The most important factor due to energy release (surface heating) is density change in the region influenced by the heating. This region structure is controlled by the convection and thermal conductivity processes. At the same time density change (diminishing due to temperature rise) will change boundary layer thickness. Situation is similar to the local flow nearby local surface distortion, but in our case effective surface distortion is created due to temperature (density) change. The difference is that the distortion shape is not known beforehand but is formed due to energy release in the boundary layer and due to the region with smaller density formation. Previous analysis of the disturbed flow nearby local surface distortions allowed to develop corresponding mathematical problems and to find distortions parameters influencing boundary layer flow [2].

In the paper corresponding asymptotical analysis was applied to analyze energy release influence on the separation of the boundary layer flow along with the influence on the boundary layer flow instability.

Investigated are local flows in the laminar boundary layer nearby heating elements located on the surface and flows near porous surface. On the basis of asymptotic analysis mathematical models are derived and similarity parameters are found. Described are unsteady local heating regimes providing boundary layer separation and laminar-turbulent transition control. Presented are numerical and analytical analysis results.

At present time intensive investigations are conducted associated with new methods of boundary layer control due to energy release. The energy source may be due to the electrical discharge, surface heating or cooling. Along these methods mechanical devices are analyzed, so called MEMS, or methods associated with suction or injection. Application of such methods allows suppressing boundary layer separation, to change boundary layer transition position along with the influence on the turbulent boundary layer parameters.

This paper is aimed at analysis of possible regimes of local surface heating to determine arising disturbances in the boundary layers. It is supposed as well to formulate corresponding mathematical problems and to determine similarity parameters. Analogous analysis is done for disturbed flows near porous surface.

It is supposed that on the streamlined surface are located parts having temperature which differs from the temperature of an ambient gas. It is supposed as well that this temperature depends on time. Practically is easy to fulfill such method of control having conducted wires or strips and using electrical current for local heating [1].

Let us consider physical aspects of such flows. The most important factor associated with the surface heating is gas density change in the region where energy release influence is significant. Dimensions of this region are determined by convection and heat conduction processes. Temperature increase will lead to the density decrease which will change boundary layer thickness. This change may induce in the external inviscid flow corresponding pressure disturbances. Situation is very similar to the flow in the boundary layer with the surface distortions. The main difference is that in the considered case geometry of surface distortions are not known beforehand but is determined by energy release and by corresponding decreased density region formation. Flow analysis in the boundary layer disturbed by an abrupt change of surface temperature and catalytic properties distribution is presented in [3-5].

For the subsequent analysis we will use the following papers [4, 6-8] results where local surface distortions located on the bottom of the surface were analyzed. It is supposed that the Reynolds number is large but doesn't exceed the critical value corresponding to the laminar-turbulent transition. Subsequent analysis is base on the derivation of estimates of possible physical mechanisms and similarity parameters determination.

Considered is the supersonic or subsonic viscous gas flow near a flat semi-infinite plate. It is supposed that the Reynolds number is large but is subcritical corresponding to the laminar flow $Re = \rho_\infty u_\infty l / \mu_\infty = \varepsilon^{-2} \gg \infty$, where ρ_∞ , u_∞ , μ_∞ - are density, longitudinal velocity and dynamical viscosity coefficient in undisturbed flow over the region where heated part of plate is located, l - is a distance from a leading edge to a zone of energy release. The next nondimensional values are chosen for the Cartesian coordinates, velocity vector components, density, pressure, dynamical viscosity coefficient

$$xl, yl, zl, lu_\infty^{-1}t, u_\infty u, u_\infty v, u_\infty w, \rho_\infty \rho, u_\infty^2 R^{-1}T, \mu_\infty \mu -.$$

In general case it is supposed that temperature change in the local region is finite $\Delta T \sim T \sim O(1)$, and the region of increased temperature is characterized by the next longitudinal size $a \leq O(1)$, next transversal size $b \leq O(1)$ and characteristic time of temperature change $O(\tau)$.

Undisturbed flow is 2-D and steady but disturbed flow is supposed to be unsteady and 3-D. Preliminary analysis is conducted for 2-D flow and then results are generalized for 3-D flows.

Following to the method of matched asymptotic expansions [8], in the beginning let's consider the region having identical sizes comparable with the body length $x \sim y \sim z \sim O(1)$. For large Reynolds number the flow in this region is described by the Euler equations. For the flat plate having zero angle of attack and zero thickness these equations solution is a solution describing undisturbed flow. To fulfill no-slip conditions it is needed to introduce boundary layer - region located nearby the surface and having the next sizes $x \sim z \sim O(1), y \sim O(\varepsilon)$.

Local surface heating may lead to the effective distortion formation, thickness of which may be evaluated using longitudinal impulse equation.

Surface temperature change will cause corresponding gas density change in the layer located nearby the surface $\Delta \rho \sim \rho \sim O(1)$. If the Prandtl number is finite $Pr \sim O(1)$ in general case it may be deduced that thickness of local viscous layer and thickness of temperature conducting layer are comparable. In the near wall layer longitudinal velocity is proportional to the distance from the wall y / ε , $u \sim O(y / \varepsilon)$. If values of convective and diffusion terms in the longitudinal momentum equation have the same order then the next estimate can be obtained for the thickness of local layer as a function of its longitudinal size a

$$y \sim O(\varepsilon a^{1/3}) \leq O(\varepsilon)$$

Subsequent analysis depends on the longitudinal size of heated part of the surface. At least three different regimes described by different mathematical models may be formulated.

The first one corresponds to the longitudinal size smaller than the boundary layer thickness. If sizes of the disturbed region have the same orders

$$a \sim O(\varepsilon a^{1/3}), \quad a \sim \varepsilon^{3/2}$$

we will get the disturbed region where the flow is described by Navier-Stokes equations with boundary conditions taking into account rarefied gas effects (slip conditions). Characteristic time in this region has the next order $\tau \sim \varepsilon^{-1}$.

The problem is presented in [13] and its solution can be found only by numerical methods.

For relatively larger sizes of the heated part $a^2 Re^{3/2} = Re_1 \gg \infty$ the disturbed flow will be described by so called equations for compensation regime [7]. This regime will exist for length comparable with the boundary layer thickness as well as for larger sizes but lesser than the length scale comparable with the so called free interaction scale. Corresponding to this regime

$$a \sim O(\varepsilon^{3/4}), \quad \Delta y \sim O(\varepsilon^{5/4} \Delta T), \quad \Delta p \sim O(\varepsilon^{1/2} \Delta T)$$

Corresponding mathematical problem may be written as follows

$$Sh_2 \frac{\partial u_b}{\partial t_b} + u_b \frac{\partial u_b}{\partial x_b} + v_b \frac{\partial u_b}{\partial y_b} + \Pi_1 w_b \frac{\partial u_b}{\partial z_b} + T_b \frac{\partial p_b}{\partial x_b} = \frac{\partial^2 u_b}{\partial y_b^2}$$

$$Sh_2 \frac{\partial w_b}{\partial t_b} + u_b \frac{\partial w_b}{\partial x_b} + v_b \frac{\partial w_b}{\partial y_b} + \Pi_1 w_b \frac{\partial w_b}{\partial z_b} + T_b \frac{\partial p_b}{\partial z_b} = \frac{\partial^2 w_b}{\partial y_b^2}$$

$$Sh_2 \frac{\partial T_b}{\partial t_b} + u_b \frac{\partial T_b}{\partial x_b} + v_b \frac{\partial T_b}{\partial y_b} + \Pi_1 w_b \frac{\partial T_b}{\partial z_b} = \frac{\partial^2 T_b}{\partial y_b^2}$$

$$\frac{\partial u_b}{\partial x_b} + \frac{\partial v_b}{\partial y_b} + \frac{\partial w_b}{\partial z_b} = 0$$

$$u_b(x_b, 0, z_b, t_b) = w_b(x_b, 0, z_b, t_b) = v_b(x_b, 0, z_b, t_b) = 0$$

$$T_b(x_b, 0, z_b, t_b) = T_w(x_b, z_b, t_b)$$

$$u_b \in y_b + d, \quad d_1 = \int_0^{\infty} (1 - T_1) d\eta + d, \quad T_1(x, \infty) \in 1, \quad y_b \in \infty$$

$$d(-\infty) \in 0,$$

$$p_b(x_b, t) = -B_2 \frac{\partial d_1}{\partial x_b}$$

This problem was numerically solved. Numerical results were obtained for the next surface temperature distribution

$$T_w(|x_b| \leq 0.5, t) = 1. + (1 - \exp(-t))(0.25 - x_b^2),$$

$$T_w(|x_b| > 0.5, t) = 1., B_2 = 1$$

On the fig.1 induced pressure distribution is presented $p_b(x_b, t \rightarrow \infty)$ for large time values.

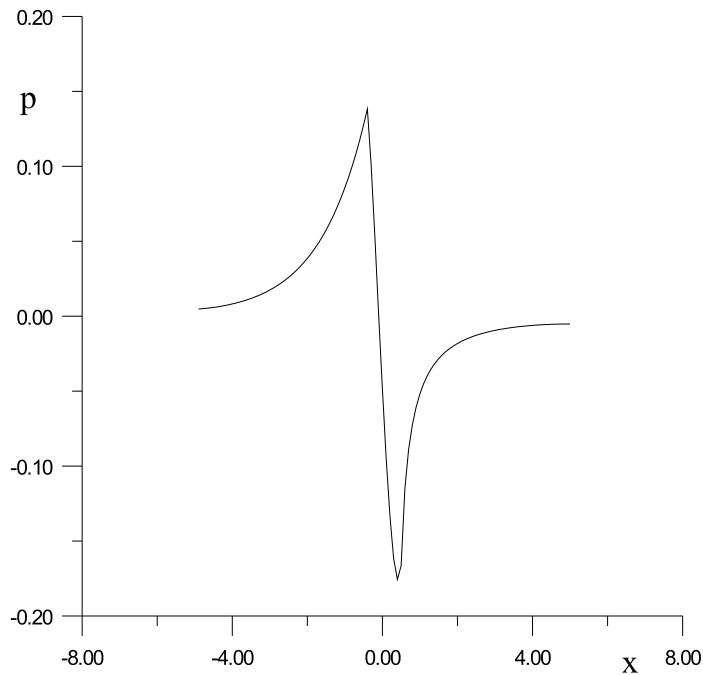


Fig.1

Flows near porous walls. Porous wall structure suppose that due to pressure difference on the external and internal sides of porous surface may lead to the distributed suction (in the regions of relatively high pressure) or distributed injection. In many cases it may be supposed that mass transfer obeys the Darcy law (or linear dependence between vertical velocity on the wall and disturbed pressure distribution).

This boundary condition allow in fact to reconsider early obtained results [9-11] describing self-induced boundary layer separation for the case of passive flow control.

Using results obtained in [9] mathematical problem for flows near porous walls may be formulated as

follows

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + \frac{\partial p}{\partial x} = \frac{\partial^2 u}{\partial y^2} \quad \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$y = 0 \quad v = -\beta p, \quad u = 0 \quad y \rightarrow \infty \quad u = y + A(x)$$

$$x \rightarrow -\infty \quad u = y \quad p = -\frac{\partial A}{\partial x}$$

This problem differs from problems describing disturbed flow near impermeable wall due to condition for the vertical velocity on the wall [9-11]. In fact such conditions are well known in fluid mechanics.

For small values of self-induced pressure next form of solution may be considered

$$u = y + u_1, \quad v = v_1, \quad p = p_1$$

This form of solution gives next form of equations for the first approximation

$$y \frac{\partial u_1}{\partial x} + v_1 + \frac{\partial p_1}{\partial x} = \frac{\partial^2 u_1}{\partial y^2}$$

$$\frac{\partial u_1}{\partial x} + \frac{\partial v_1}{\partial y} = 0$$

As usual solution may be presented in the normal mode approximation

$$(u_1, v_1, p_1, A) = e^{\alpha x} (U, V, P, B) \quad y \alpha U + V + \alpha P = U'' \quad U(\infty) = B$$

$$y \alpha U' = U'''$$

After some transformations we will get Airy equation

$$y = \alpha^{-1/3} Y \quad U' = F \quad YF = F''$$

With the next solution

$$F = C Ai(Y) \quad F = \frac{\partial U}{\partial y} = \alpha^{1/3} \frac{\partial U}{\partial Y} \quad F = \frac{\partial U}{\partial Y} = C \alpha^{-1/3} Ai(Y)$$

$$U(\infty) = \int_0^{\infty} F dY = C\alpha^{-1/3} \int_0^{\infty} Ai(Y) dY = \frac{C}{3\alpha^{1/3}} \quad P = -\alpha B = -\frac{C\alpha^{2/3}}{3}$$

$$(\alpha - \beta)P = \alpha^{2/3} U_w'' = C\alpha^{1/3} Ai'(0)$$

$$-\frac{C\alpha^{2/3}}{3}(\alpha - \beta) = C\alpha^{1/3} Ai'(0)$$

Introducing new variables

$$\alpha^{1/3}(\alpha - \beta) = -3Ai'(0) \quad \alpha = \delta^3 \beta$$

Eventually we will get next relation associating increment of growth and wall velocity parameter β

$$\delta(\delta^3 - 1) = -\frac{3Ai'(0)}{\beta^{4/3}}$$

Next two limits can be considered

The first one corresponding small porosity $\beta \ll 0$

$$\delta \approx \left[-\frac{3Ai'(0)}{\beta^{4/3}} \right]^{1/4} \approx \frac{1}{\beta^{1/3}} \left[-3Ai'(0) \right]^{1/4}$$

Then since $\alpha = \delta^3 \beta$

we will get

$$\alpha \approx \left[-3Ai'(0) \right]^{3/4}$$

The same result as was obtained in classical case (impermeable plate) [9]

In the second limit-corresponding to large porosity $\beta \gg \infty$

$$\delta \approx 1, \quad \alpha \approx \beta$$

So, large α values correspond to small length of disturbed region.

It is important that new boundary condition on the wall is linear in the approximation considered so mathematical problem for linear regimes may be considered as a uniform one.

To investigate stability problems is needed to consider unsteady mass transfer regimes.

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