

Space transportation challenges

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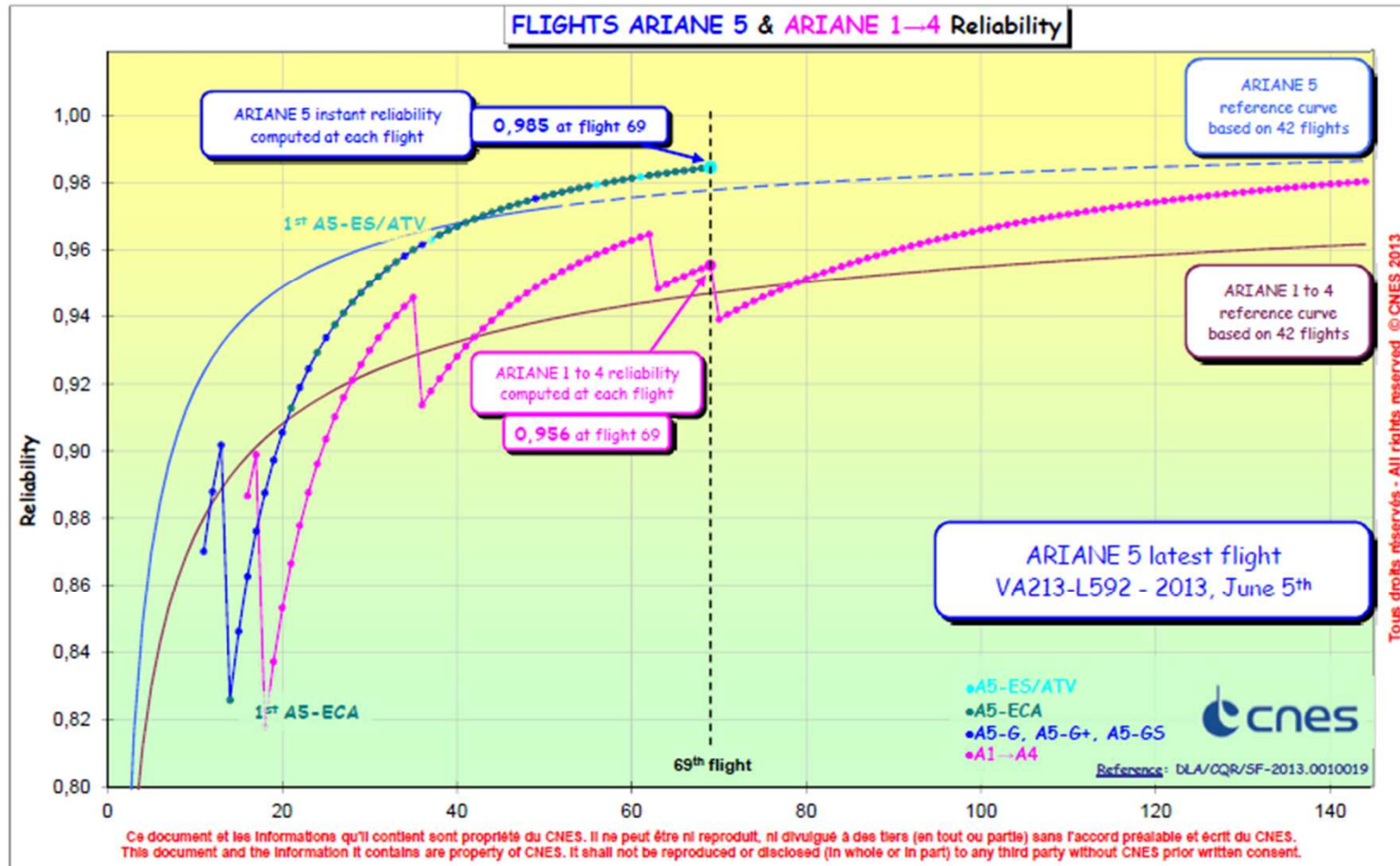
5th EUCASS July 1st 2013 - Munich

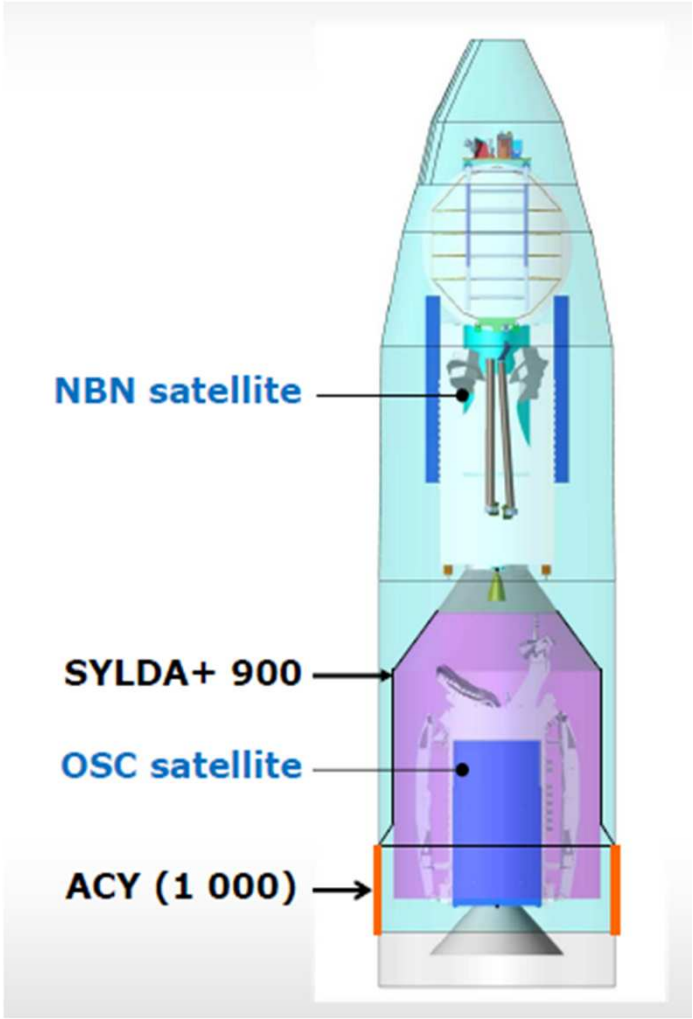
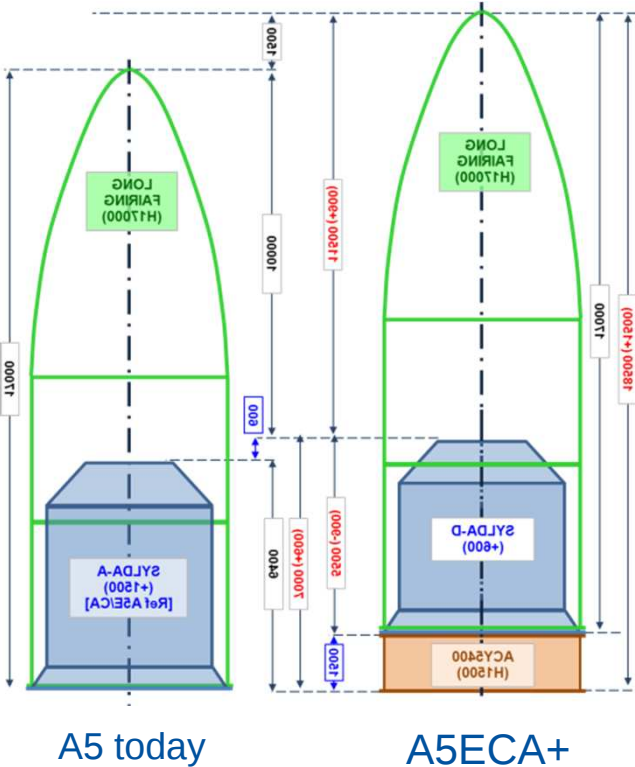
ARIANE, SOYOUZ and VEGA

- European family of launchers Ariane, Soyouz and Vega is operational a wide range of orbits and performances achievable
- Challenges for the short term are:
 - To gain maturity for the new systems
 - To keep quality and reliability
 - To gain flexibility in operations
 - To cope with satellites evolutions (mass, volume, missions)



RELIABILITY





COMPETITIVE ENVIRONMENT IS CHANGING



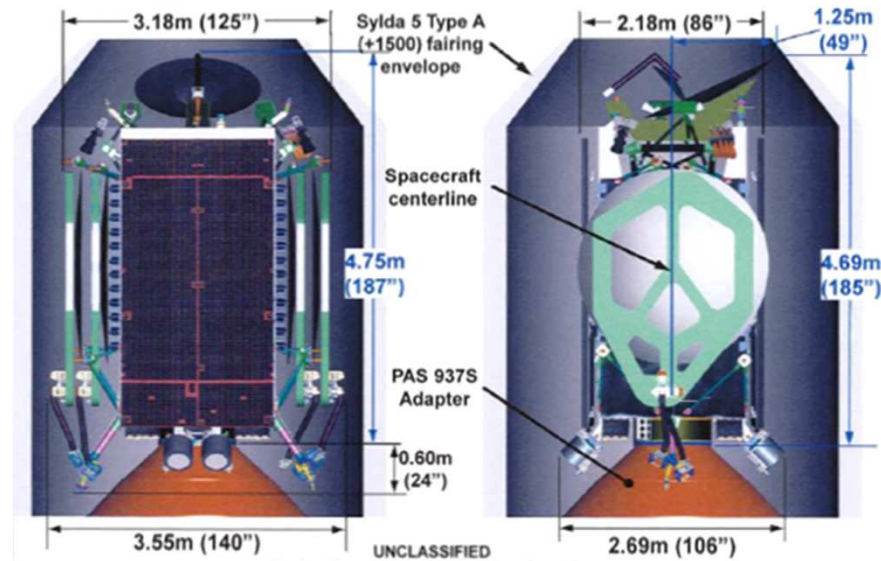
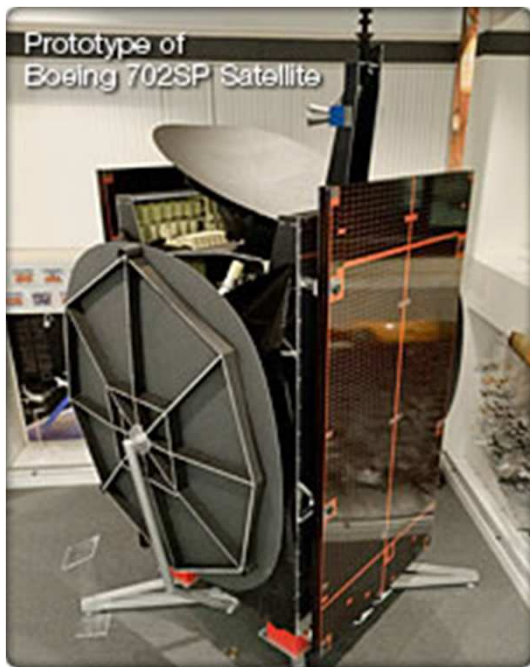
Proton

Falcon

- To anticipate the impact of environment's evolution => Strong competition
 - Proton in the segment of heavy satellites (6T class)
 - “Falcon is coming” in the segment of small medium satellites (3/4 T class)

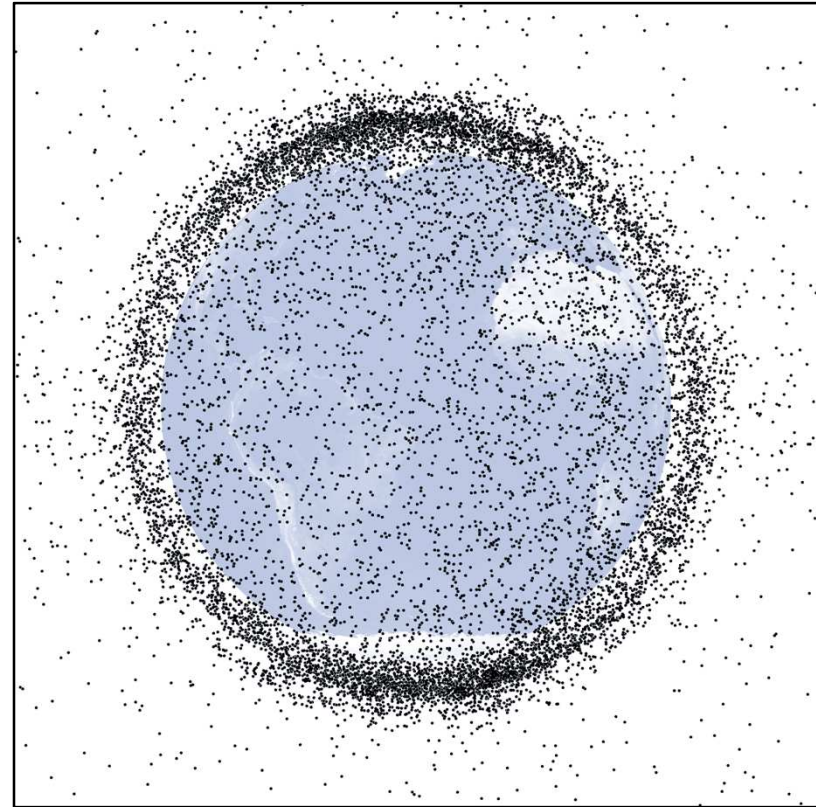
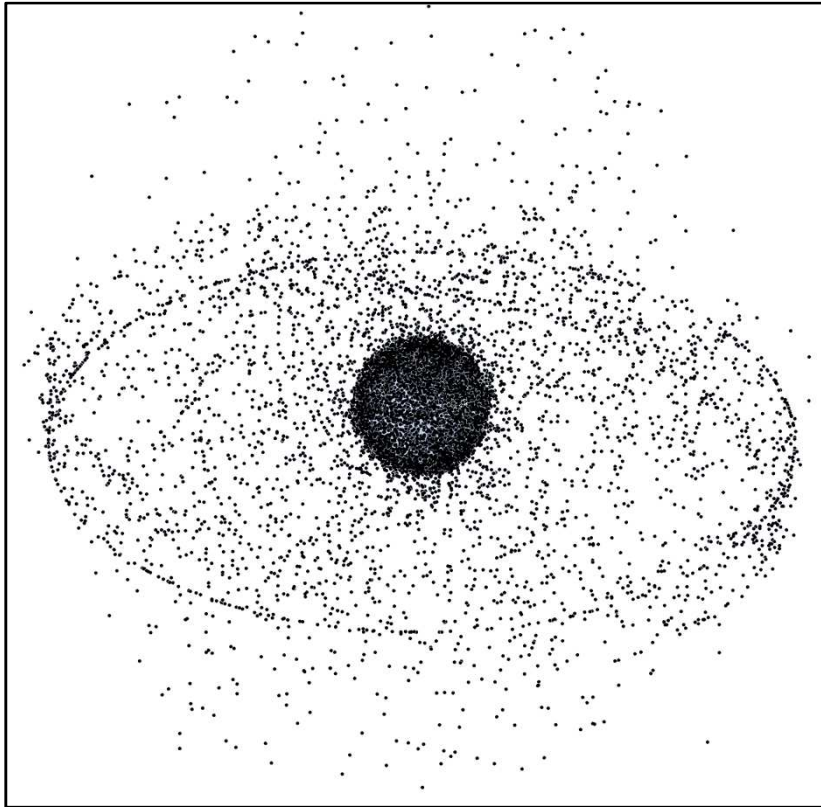
ELECTRIC PROPULSION

- A new type of satellites and missions are emerging (electric propulsion)



SPACE DEBRIS

- To cope with international and national regulations for space debris.



EUROPEAN LAUNCHER MAIN ORIENTATIONS

- Two main developments running in parallel:
 - A5ME with its re-ignitable upper stage (horizon 2018)
 - Ariane 6 based on a PPH configuration (horizon 2021) and commonalities on upper cryo stage with A5ME.



A5ME



A6 PPH

ARIANE 6 DESIGN DRIVERS



- 7 tons GTO performance
- 7 years development and qualification
- 70 M€ for the launch price

- High energy at lift off and high speed to cross atmosphere
High level of reliability and availability, and low cost

Solid propulsion

- High performance and accuracy to reach the orbit
 - Customization of the mission
 - De-orbiting after the mission

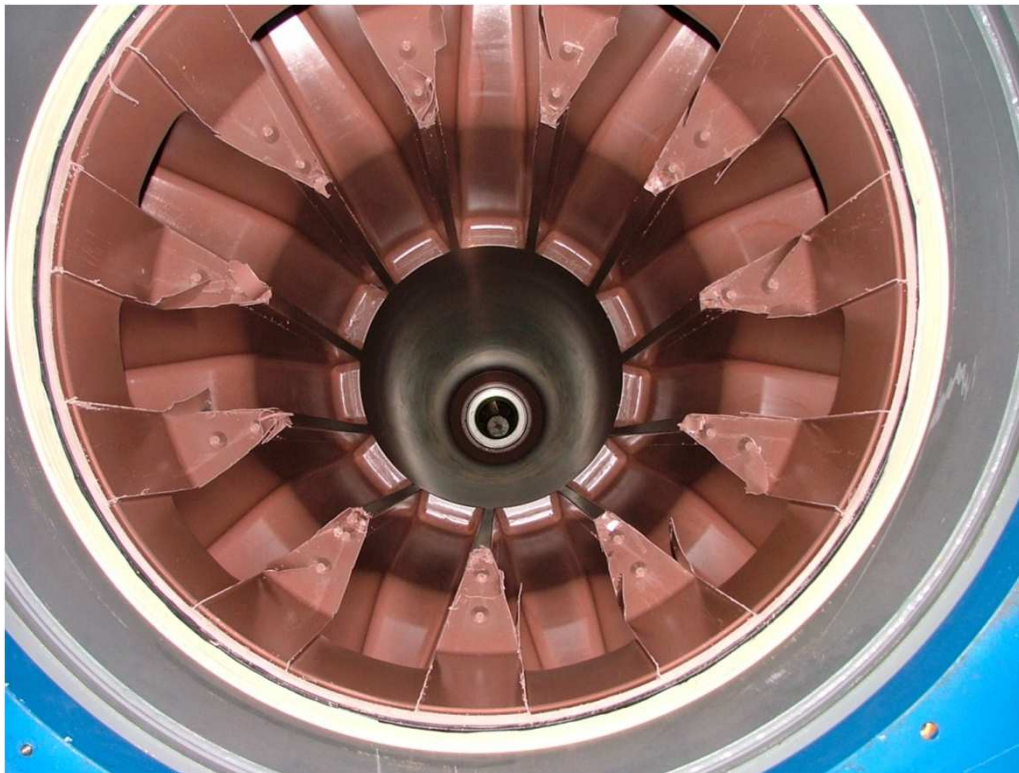
Cryogenic propulsion

HOW TO REACH THEM?

- Simple design
 - Limited number of interfaces and subsystems
 - Cadence effect on both launch system and subsystems
 - Mastery of the design by industrial primes
- 
- Creation of industrial clusters around coherent skills, technologies and processes
 - “Plug and play” approach for operations

AND ALSO ...

- To maximize the use of proven technologies, past investments on subsystems (Vinci engine) and production facilities (casting pits...)



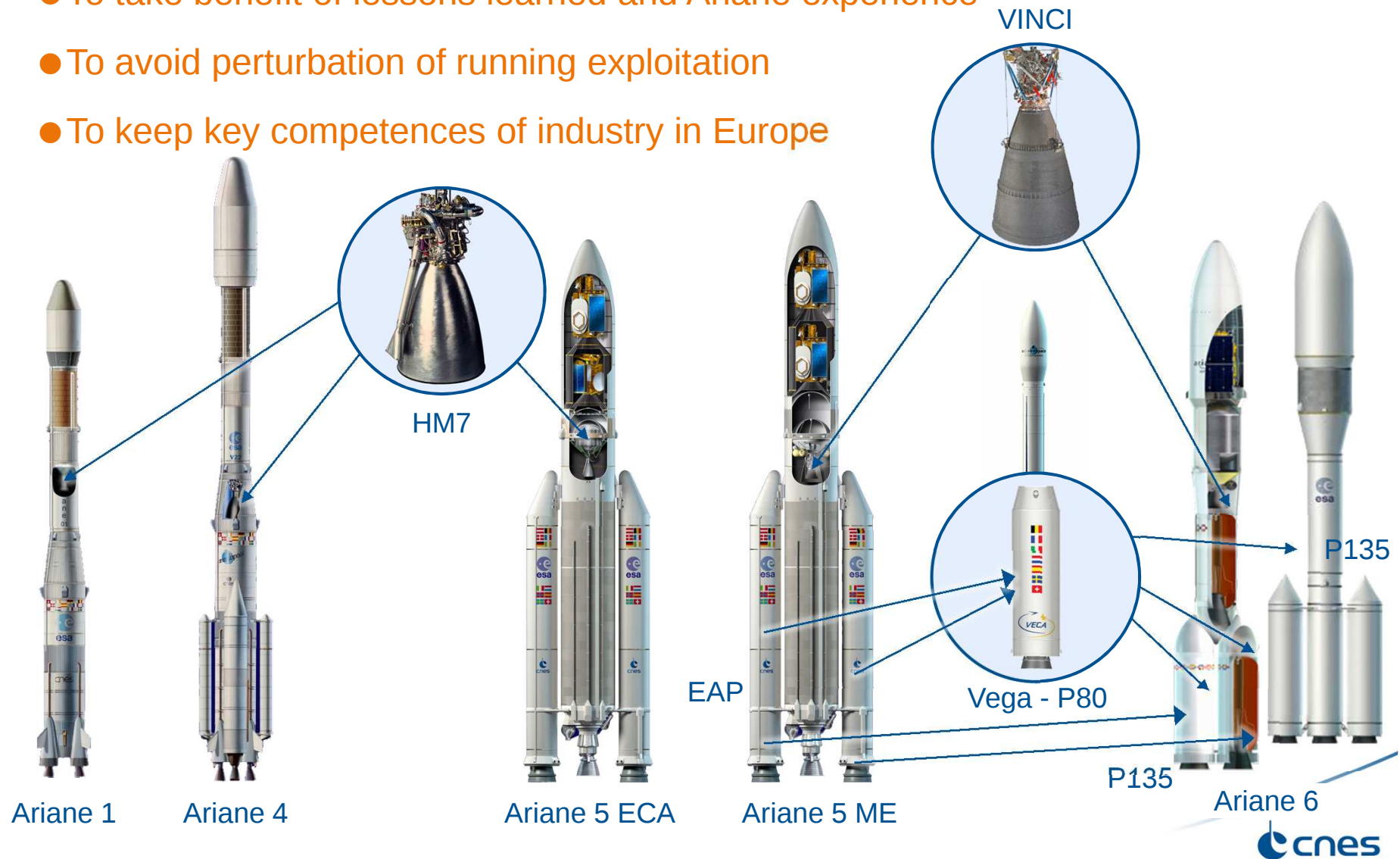
P80 – DM Loaded Motor Case



Vinci Engine test
campaign 2013

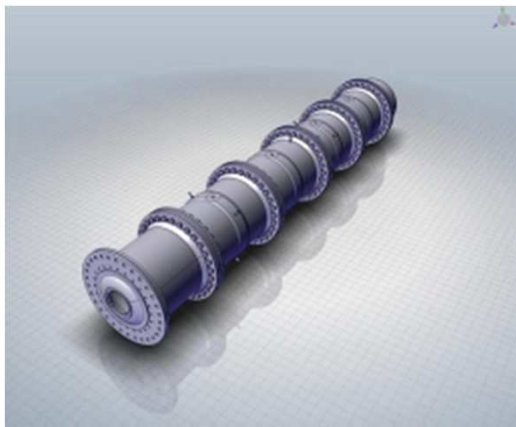
AND ALSO ...

- To take benefit of lessons learned and Ariane experience
- To avoid perturbation of running exploitation
- To keep key competences of industry in Europe

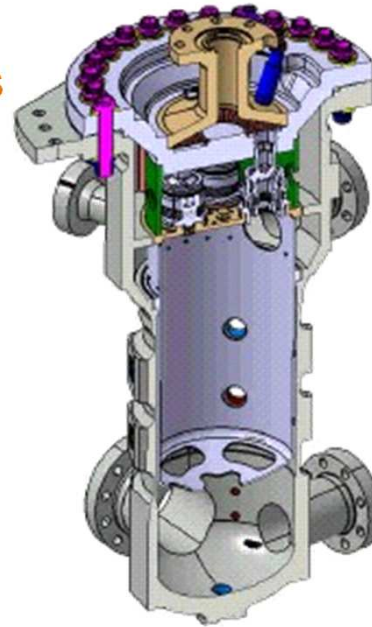


INNOVATION, SCIENCE AND RESEARCH

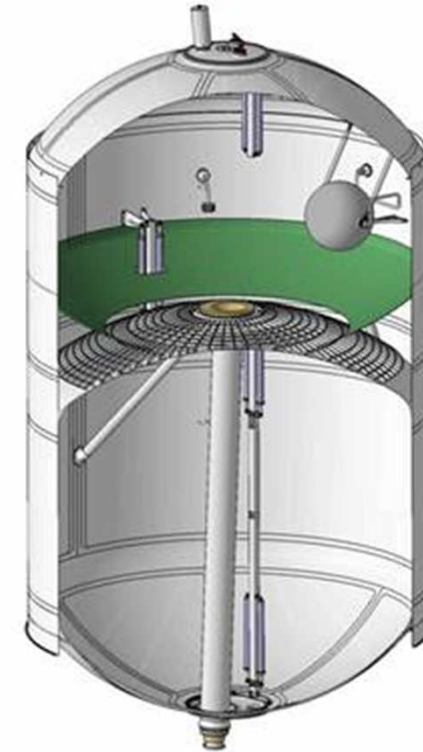
- Management of cryogenic propellant behavior during ballistic phases
- Mastery of thrust oscillations on solid propulsion
- Control of transient phases (system, engines,...)
- High frequency behavior of liquid propulsion engines
- Flight control and guidance
- Acoustic design of launch facilities



Demo POD-X



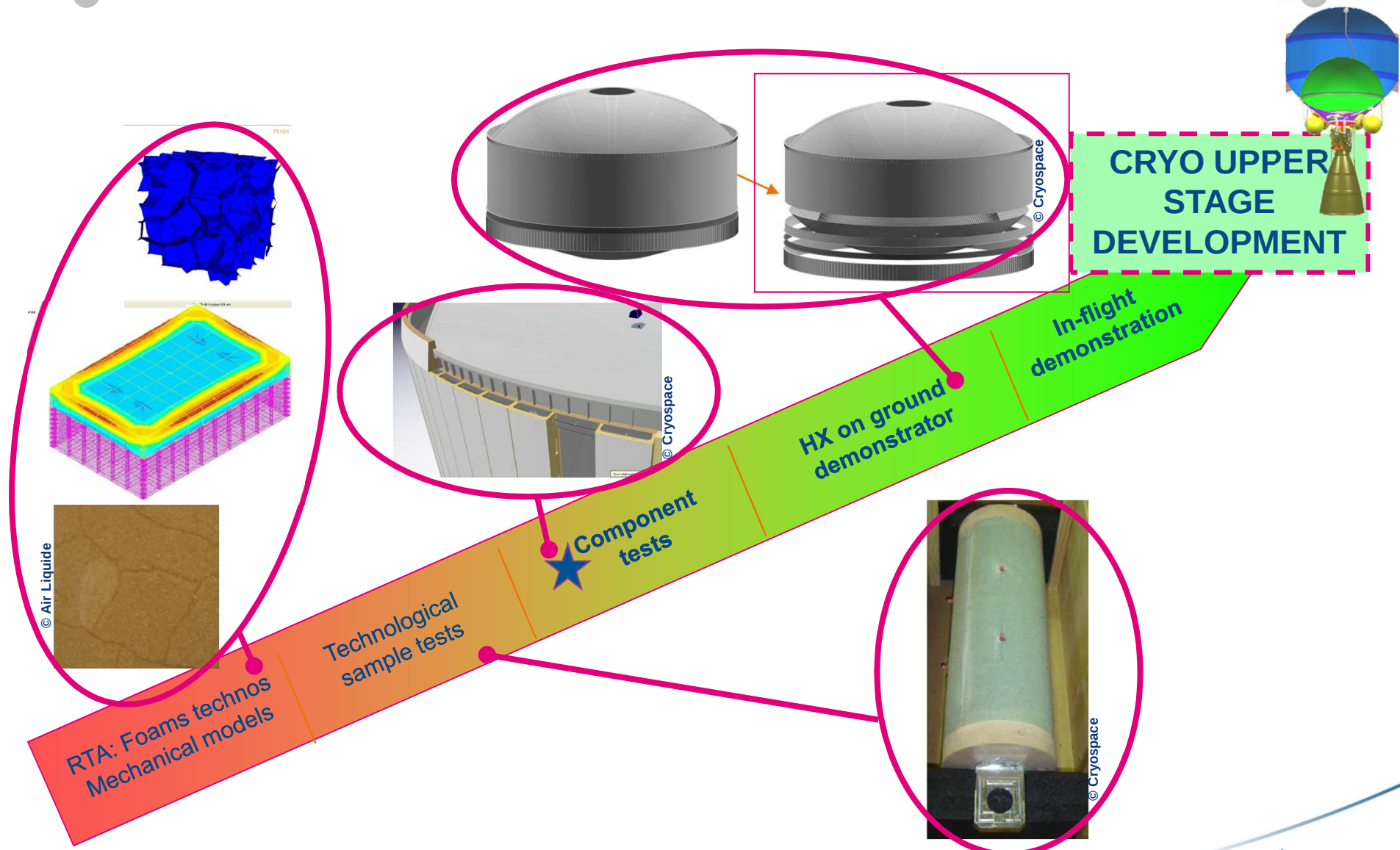
Vulcain X



HX

Research and Technology – for cryo stages

Example: COLD THERMAL PROTECTION MATURATION

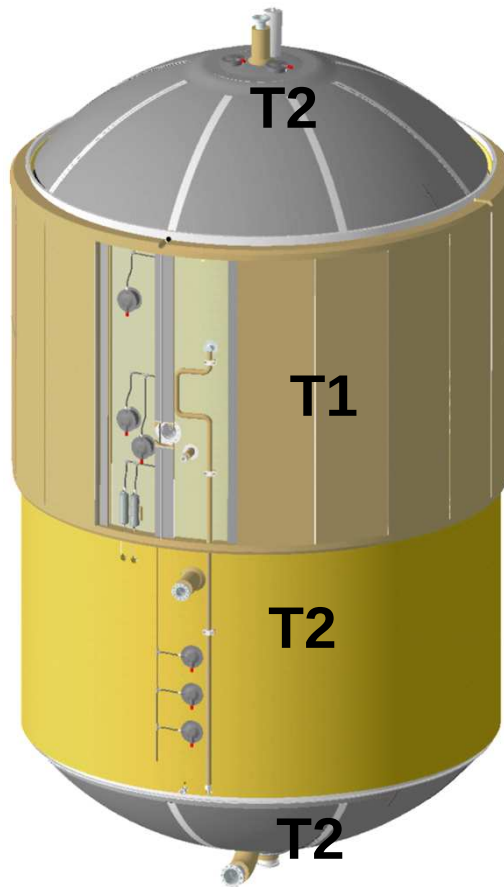


Technological Demonstrators for A5 and A6: H-X

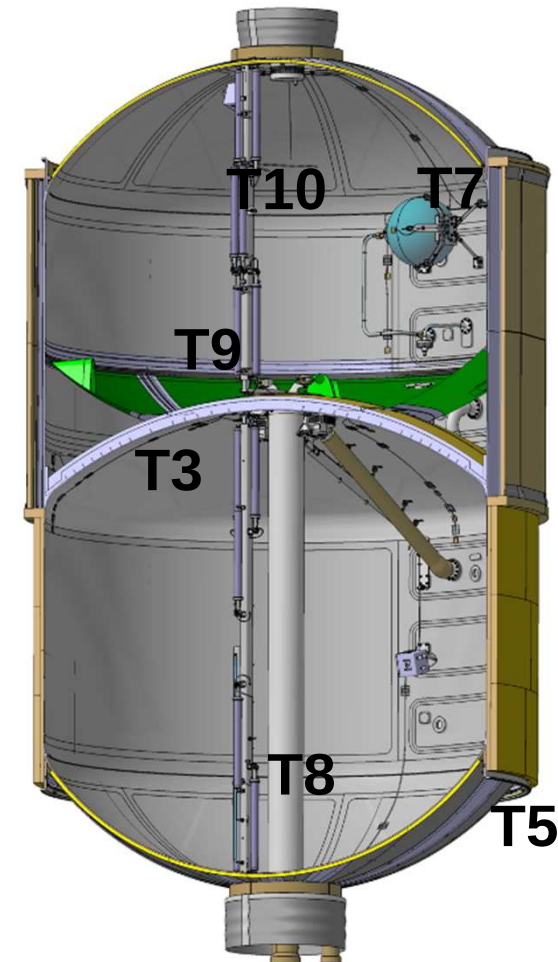
HXG-Technologies demonstration



TRL 6 achieved for



- T1 : Foam-MLI-Foam (FMLIF)
- T2 : Combined TP open cells
- T3 : Insulation for ortho-grid Cl
- T5 : Vapour cooling
- T7 : Pressurisation 20K
- T8 : Internal line
- T9 : Anti-wetting device
- T10 : Level gauges + cameras



Technological Demonstrators for A6

SOLID PROPULSION : EXTRA LARGE CASING DEMO

SRM PROJECT : Insulated Case Demo.

→ **Program** : design, manufacture and test mechanically a large and representative insulated case

→ **Main objectives**

- Demonstrate a low cost manufacturing process of the whole insulated case
- Validate the capability to withstand high mechanical fluxes on skirts
- Validate the technology of the large mandrel for the insulated case

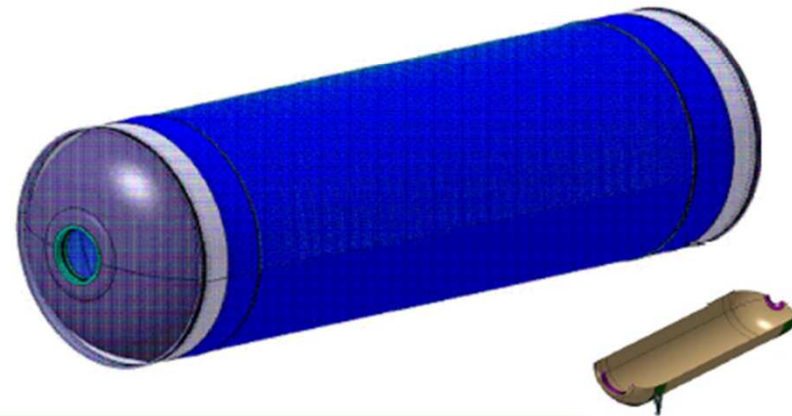


→ **Main 'Demo' characteristics**

- Scale one feasibility diameter : 3.7 m
- Nominal maximum pressure : 11 Mpa
- Nominal axial flux : 1500 N/mm

→ **Industrial Organization :**

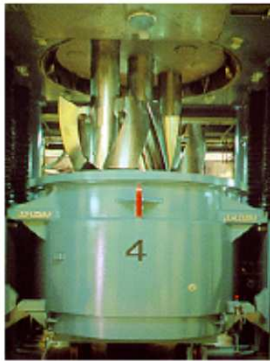
- Astrium : case, skirts and tests
- Herakles : thermal insulation



Technological Demonstrators for A6

SOLID PROPULSION : BI-SCREW NEW PROCESS

SRM : Continuous Mixing and Casting Demo.



From batch mixers
to
continuous twin screw

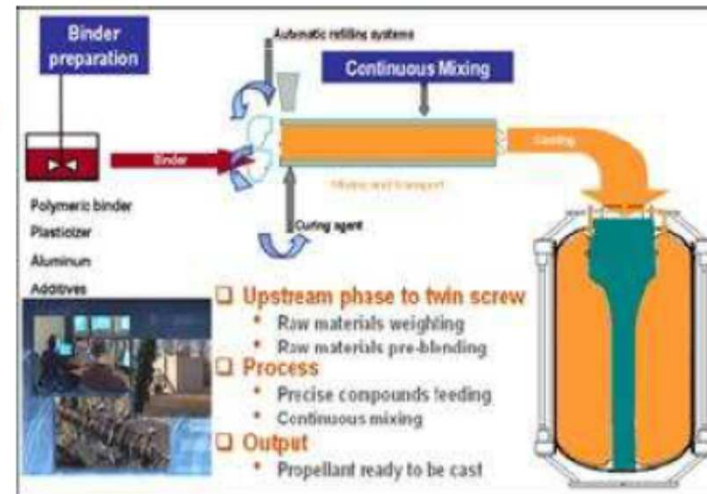


→ Objectives : reach TRL 5/6 for P180

- Low cost alternative to the “batch” mixing
- Produce an ‘equivalent’ propellant

→ Program and activities :

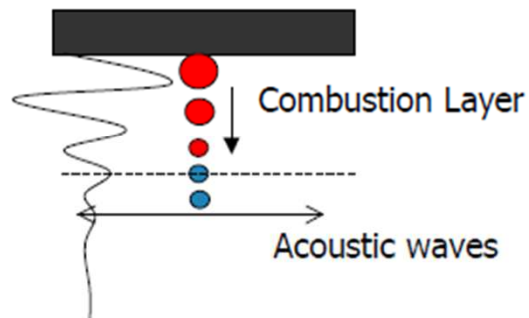
- Test at pilot scale located at ESM
- Validate process parameters influent on propellant behavior
- Demonstrate the process robustness
- Define scale 1 means and plant



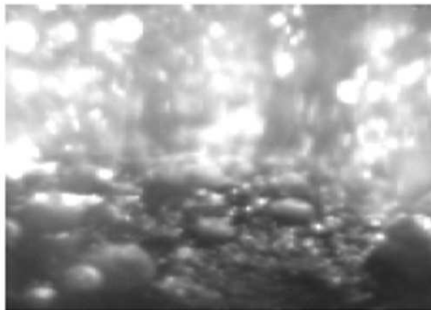
Technological Demonstrators for A6

SOLID PROPULSION : Combustion Instabilities Demo

SRM PROJECT : Pressure Oscillation Demo.



Axial acoustic waves coupling with radial combustion layer can generates some pressure oscillation phenomenon's

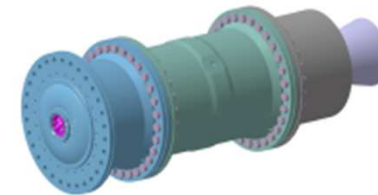


→ Program : select a propellant formulation and test it at subscale level

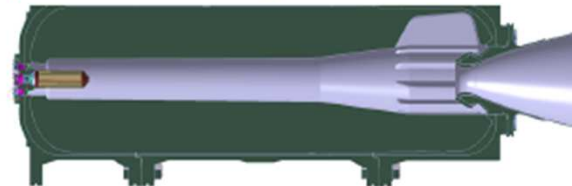
- Focusing on ITHAC pressure oscillation
- Confirming that Vortex Shedding are not influent

→ Selected reduced scale

- Propellant mass 1.1 T
- case length 2.2 m
- Diameter 0,7 m



→ First planned firing test in 2014



CONCLUSION

- Science and technology innovations are the key elements for an independent European access to space
- Optimization of industrial processes and operations are the key issues of A6 to reach low costs

